IRD CLASS FROM RTM

paragraph	segment_a	req_title	text	Add	New	Clarification
_id	llocation			Text		
AM1-	FOS CSMS	AM-1	The EOC shall have the			
0020		Spacecraft	capability to send (via			
		Command	EDOS/Ecom and the SN, GN,			
		S	DSN, or WOTS) and the AM-			
			1 spacecraft shall have the			
			capability to receive			
			spacecraft commands in			
			CCSDS CLTUs (as defined in			
			AM-1 ICD 106).			
AM1-	FOS CSMS	AM-1	The EOC shall have the			
0030		Instrumen	capability to send (via			
		t	EDOS/Ecom and the SN, GN,			
		Command	DSN, or WOTS) and the AM-			
		S	1 spacecraft shall have the			
			capability to receive			
			instrument commands in			
			CCSDS CLTUs (as defined in			
			AM-1 ICD 106).			

FOS CSMS	AM 1 Real	The AM 1 spacecraft shall	
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		1 ICD 106) via EDOS/Ecom	
		and the SN, GN, DSN, or	
		WOTS interfaces.	
FOS CSMS	AM-1	The AM-1 spacecraft shall	
	Recorded	have the capability to send	
	H/K	(in CADU format) and the	
	Telemetry	EOC shall have the	
	-	capability to receive (in	
		EDUs containing CCSDS	
		l	
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		and instrument	
		housekeeping telemetry	
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		WOTS interfaces.	
		FOS CSMS AM-1 Recorded H/K	Telemetry (in CADU format) and the EOC shall have the capability to receive (in EDUs containing CCSDS telemetry packets and CLCWs) real time AM-1 spacecraft and instrument housekeeping telemetry packets (as defined in AM-1 ICD 106) via EDOS/Ecom and the SN, GN, DSN, or WOTS interfaces. FOS CSMS AM-1 Recorded H/K Telemetry EOC shall have the capability to send (in CADU format) and the EOC shall have the capability to receive (in EDUs containing CCSDS telemetry packets) recorded AM-1 spacecraft and instrument housekeeping telemetry packets (as defined in AM-1 ICD 106) via EDOS/Ecom and the SN, GN, DSN, or

AM1-	FOS CSMS	AM 1	The AM-1 spacecraft shall	
0090	TOS CSIVIS			
0090		Dump	have the capability to send	
		Telemetry	(in CADU format) and the	
			EOC shall have the	
			capability to receive (in	
			EDUs containing CCSDS	
			telemetry packets and	
			CLCWs) AM-1 SCC, CTIU,	
			and instrument	
			microprocessor memory	
			dump telemetry packets	
			(as defined in AM-1 ICD	
			106) via EDOS/Ecom and	
			the SN, GN, DSN, or WOTS	
			interfaces.	
AM1-	FOS CSMS	Pre-	The EOC shall have the	
0120	'	Launch	capability to send and the	
		AM-1 S/C	AM-1 spacecraft shall have	
		Command	the capability to receive	
		S	spacecraft commands in	
			CCSDS CLTUs (as defined in	
			AM-1 ICD 106) via pre-	
			launch test configurations	
			which include the AM-1	
			Spacecraft Checkout	
			Station, Ecom, and EDOS or	
			ETS.	
			E13.	

AM1-	FOS CSMS	Pre-	The AM-1 spacecraft shall	
0125		launch	have the capability to send	
		R/T H/K	(in CADU format) and the	
		Telemetry	EOC shall have the	
		-	capability to receive (in	
			EDUs containing CCSDS	
			telemetry packets and	
			CLCWs) real time AM-1	
			housekeeping telemetry	
			packets (as defined in AM-	
			1 ICD 106) via pre-launch	
			test configurations which	
			include the AM-1	
			Spacecraft Checkout	
			Station, Ecom, and EDOS or	
			ETS.	

AM1-	FOS CSMS	Pre-	The AM-1 spacecraft shall	
0130		launch	have the capability to send	
		Recorded	(in CADU format) and the	
		H/K	EOC shall have the	
		Telemetry	capability to receive (in	
		-	EDUs containing CCSDS	
			telemetry packets and	
			CLCWs) recorded AM-1	
			housekeeping telemetry	
			packets (as defined in AM-	
			1 ICD 106) via pre-launch	
			test configurations which	
			include the AM-1	
			Spacecraft Checkout	
			Station, Ecom, and EDOS or	
			ETS.	

AM1-	FOS CSMS	Pre-	The AM-1 spacecraft shall	
0135		launch	have the capability to send	
		Dump	(in CADU format) and the	
		Telemetry	EOC shall have the	
			capability to receive (in	
			EDUs containing CCSDS	
			telemetry packets and	
			CLCWs) AM-1 SCC, CTIU,	
			and instrument	
			microprocessor memory	
			dump telemetry packets	
			(as defined in AM-1 ICD	
			106) via pre-launch test	
			configurations which	
			include the AM-1	
			Spacecraft Checkout	
			Station, Ecom, and EDOS or	
			ETS.	
AM1-	FOS CSMS	Launch	The SCS shall have the	
0140		Telemetry	capability to send (in CADU	
			format) and the EOC shall	
			have the capability to	
			receive (in EDUs containing	
			CCSDS telemetry packets)	
			AM-1 spacecraft telemetry	
			data (as defined in AM-1	
			ICD-106) during spacecraft	
			launch via launch	
			configurations which	
			include EDOS and Ecom.	

AM1-	FOS	SSIM	The EOC shall have the	
0150		Commandi	capability to send and the	
		ng	SSIM shall have the	
			capability to receive AM-1	
			spacecraft and instrument	
			commands in CCSDS CLTU	
			format (as defined in AM-1	
			ICD-106).	
AM1-	FOS+CSMS	SSIM R/T	The SSIM shall have the	The SSIM outputs
0160		H/K	capability to send and the	telemetry in CADU
		Telemetry	EOC shall have the	format. The EOC
			capability to receive (in	receives the
			EDUs containing CCSDS	telemetry as CCSDS
			telemetry packets)	packets and CLCWs.
			simulated real time AM-1	CADU-to-CCSDS
			spacecraft and instrument	packet conversion is
			housekeeping telemetry	performed by the
			packets and Command Link	ETS.
			Control Words (as defined	
			in AM-1 ICD-106).	

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AM1-	FOS+CSMS	SSIM	The SSIM shall have the	The SSIM outputs
0170		Recorded	capability to send and the	telemetry in CADU
		H/K	EOC shall have the	format. The EOC
		Telemetry	capability to receive (in	receives the
			EDUs containing CCSDS	telemetry as CCSDS
			telemetry packets)	packets and CLCWs.
			simulated recorded AM-1	CADU-to-CCSDS
			spacecraft and instrument	packet conversion is
			housekeeping telemetry	performed by the
			packets (as defined in AM-	ETS.
			1 ICD-106).	
AM1-	FOS+CSMS	SSIM	The SSIM shall have the	The SSIM outputs
0200		Dump	capability to send and the	telemetry in CADU
		Telemetry	EOC shall have the	format. The EOC
			capability to receive (in	receives the
			EDUs containing CCSDS	telemetry as CCSDS
			telemetry packets)	packets and CLCWs.
			simulated AM-1 SCC, CTIU,	CADU-to-CCSDS
			and instrument	packet conversion is
			microprocessor memory	performed by the
			dump telemetry (as	ETS.
			defined in AM-1 ICD-106).	

AM1-	FOS+CSMS	Project	The AM-1 spacecraft	
0215	·	Data Base	vendor shall have the	
		from AM-	capability to provide and	
		1 Vendor	the EOC shall have the	
			capability to receive, AM-1	
			project data base	
			information containing	
			both spacecraft and	
			instrument parameters.	
AM1-	CSMS	IST	The ECS shall have the	
0220		Toolkit	capability to provide and	
		Delivery	the MISR, MOPITT, MODIS,	
			and CERES PIs/TLs shall	
			have the capability to	
			receive IST toolkit	
			software, IST toolkit	
			software upgrades, and IST	
			toolkit documentation.	
AM1-	FOS	AM-1	The AM-1 spacecraft	
0225		Spacecraft	vendor shall have the	
		Analysis	capability to provide and	
		Software	ECS shall have the	
			capability to receive	
			spacecraft analysis tools for	
			implementation and	
			integration into the EOC.	

AM1- 0230	FOS	IST Toolkit Data Import	The IST toolkit shall have the capability to accept data from a science computing facility that supports PI/TL operations, which include the following data (at a minimum): a.instrument microprocessor memory loads. b.changes in the instrument parameters	
AM1-	FOS	IST	The IST toolkit shall have	
0240		Toolkit data export	the capability to provide data to a science computing facility that supports PI/TL instrument operations, which include the following data (at a minimum): a.Microprocessor memory dumps b.Instrument analysis results	
AM1- 0270	FOS+CSMS	Software Updates	The AM-1 SDVF shall have the capability to send and ECS shall have the	
		from SDVF	capability to receive AM-1 SCC flight software updates.	

AM1- 0280	FOS+CSMS	Flight Software Dumps to SDVF	ECS shall have the capability to send and the AM-1 SDVF shall have the capability to receive AM-1 SCC flight software dumps.	
AM1- 0310	NONE	Training to AM-1 Vendor	The ECS contractor shall provide and the AM-1 spacecraft vendor shall receive training on operations of the FOS.	
AM1- 0315	NONE	Training to AM-1 PI/TLs	The ECS contractor shall provide and the AM-1 instrument teams shall receive training on operations of the IST toolkit.	
AM1- 0320	NONE	Training from AM- 1 Vendor	The AM-1 spacecraft vendor shall provide and the ECS contractor shall receive AM-1 spacecraft operations training.	
AM1- 0330	NONE	Training from AM- 1 PI/TL	The AM-1 instrument teams shall provide and the ECS contractor shall receive AM-1 instrument operations training.	

AM1-	FOS	Document	The AM-1 project shall	
0340	100		have the capability to	
		AM-1	provide and ECS shall have	
			^	
		Project	the capability to accept and	
			store AM-1 spacecraft and	
			instrument hardware and	
			software technical	
			documentation.	
AM1-	FOS CSMS	ECS RMA	ECS functions shall have an	
1000			operational availability	
			(computed as defined in	
			the Functional and	
			Performance Requirements	
			Specification for the ECS) of	
			0.96 at a minimum and a	
			mean down time (MDT) of	
			four (4) hours or less,	
			unless otherwise specified.	

AM1- 1010	FOS	RMA- Critical	The ECS FOS shall have an operational availability of	
		R/T	0.9998 at a minimum and a	
		Functions	MDT of one (1) minute or	
			less for critical real time	
			functions that support:	
			a. Launch	
			b. Early orbit checkout	
			c. Disposal	
			d. Orbit adjustment	
			e. Anomaly	
			investigation	
			f. Recovery from safe	
			mode	
			g. Routine real time	
			commanding and	
			associated monitoring for	
			spacecraft and instrument	
			health and safety	
AM1-	FOS	RMA for	The ECS FOS shall have an	
1020		non-	operational availability of	
		critical	0.99925 at a minimum and	
		R/T	a MDT of five (5) minutes	
		functions	or less for non-critical real	
			time functions.	

AM1-	FOS CSMS	AM-1	The EOC shall support	
1050		Uplink	several uplink rates to the	
		Rates	spacecraft, which include at	
			a minimum the following:	
			a.1 0 kilobits per second	
			(kbps) (SSAuplink)	
			b. 1 kbps (S-band MA	
			uplink)	
			c. 125 bits per second	
			(bps) (SSA uplink during	
			contingency operations)	
			d. 2 kbps (emergency	
			operations via S-band DSN	
			link)	
AM1-	FOS CSMS	Simultane	The EOC shall be capable of	
1060		ous	simultaneously receiving	
		Telemetry	all AM-1 telemetry data	
		Types	types.	
AM1-	FOS CSMS		The EOC shall provide the	
1070		16 kbps	capability to receive and	
		data	process real-time data	
		streams	received as two 16 kbps	
			data streams.	
AM1-	FOS CSMS		The EOC shall provide the	
1080		Recorder	capability to receive and	
		data up to	record spacecraft recorder	
		1.544	data at rates up to 1.544	
		Mbps	Mbps.	

AM1- 1090	FOS CSMS	SSIM Command Rates	The EOC shall be capable of providing CLTUs to the SSIM at the following data rates: a. 125 bps b. 1 kbps c. 2 kbps d. 10 kbps	
AM1-	FOS CSMS	Two	The EOC shall be capable of	The SSIM outputs
1100		16kbps	receiving two	telemetry in CADU
		streams	housekeeping telemetry	format. The EOC
		from SSIM	packet streams of 16 kbps	receives the
			from the SSIM.	telemetry as CCSDS
				packets and CLCWs.
				CADU-to-CCSDS
				packet conversion is
				performed by the
1.2.54	70010010			ETS.
AM1-	FOS CSMS	_	The EOC shall be capable of	The SSIM outputs
1110		telemetry	receiving a health and	telemetry in CADU
		from SSIM	safety telemetry packet	format. The EOC
			stream from the SSIM at 1	receives the
			kbps.	telemetry as CCSDS
				packets and CLCWs.
				CADU-to-CCSDS
				packet conversion is
				performed by the
				ETS.

AM1- 1120	FOS CSMS	diagnostic s/dump	The EOC shall be capable of receiving a diagnostic telemetry/memory dump packet stream from the SSIM at 16 kbps.	The SSIM outputs telemetry in CADU format. The EOC receives the telemetry as CCSDS packets and CLCWs. CADU-to-CCSDS
AM1- 1130	FOS CSMS	SSIM Recorder Dumps	The EOC shall be capable of receiving a spacecraft recorder housekeeping	packet conversion is performed by the ETS. The SSIM outputs telemetry in CADU format. The EOC
		256/512 kbps	telemetry packet stream from the SSIM at 256 kbps	receives the telemetry as CCSDS
			or 512 kbps.	packets and CLCWs. CADU-to-CCSDS packet conversion is performed by the ETS.

AM1-	FOS CSMS	Loop	ECS shall contribute a loop
1150		Delay-	delay of not greater than
		Emergenc	2.5 seconds of the total
		y R/T	system delay of five (5)
		Command	seconds for emergency
		S	real-time commands, not
			including the time needed
			for command execution.
			The loop delay is measured
			from the originator to the
			spacecraft/instrument and
			back and only applies
			when a Tracking and Data
			Relay Satellite System
			(TDRSS) link is available for
			contact to the spacecraft.